

Automatic Control

If you have a smart project, you can say "I'm an engineer"

Lecture 4

Staff boarder

Dr. Mohamed Saber Sokar

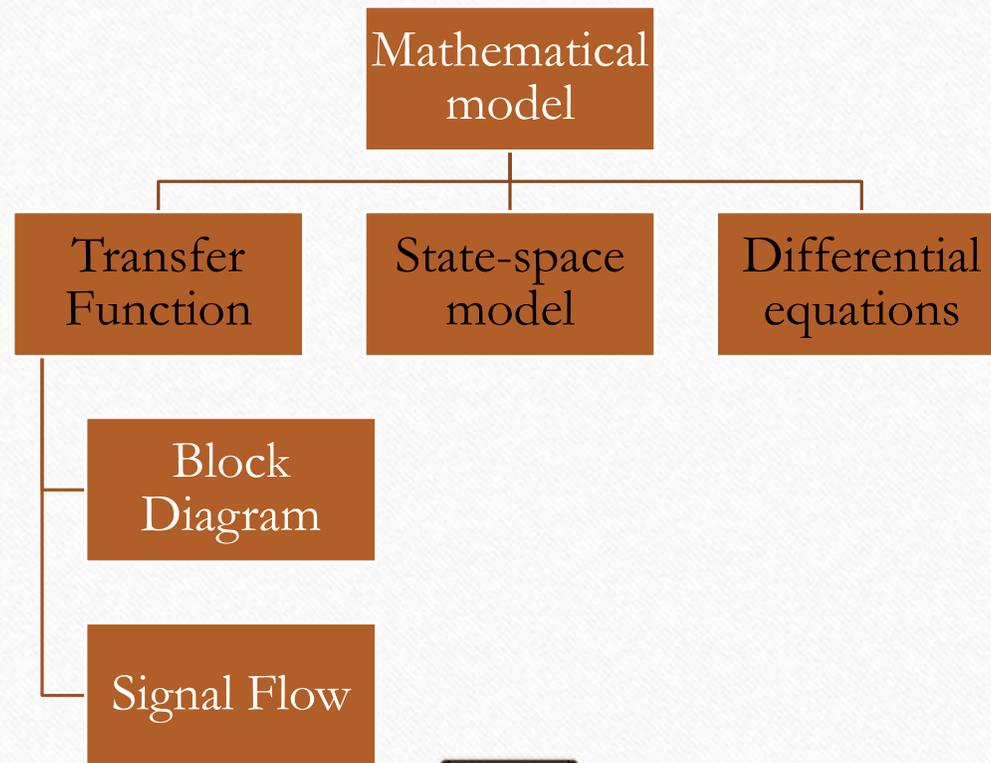
Dr. Mostafa Elsayed Abdelmonem

Automatic Control

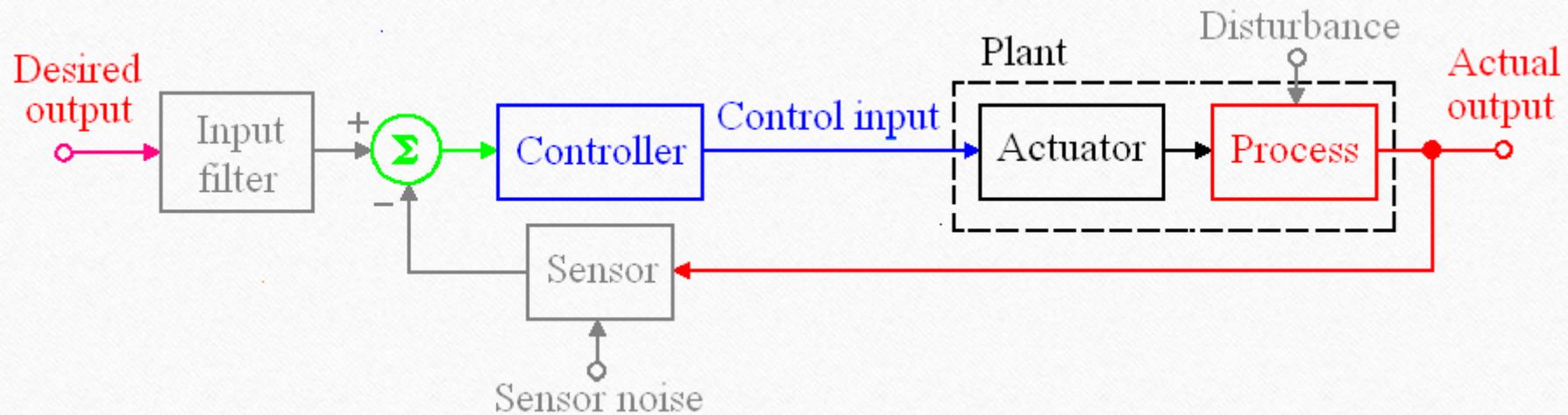
MPE 424

- **Lecture aims:**
 - Understand the Block reduction techniques
 - Identify the transfer function

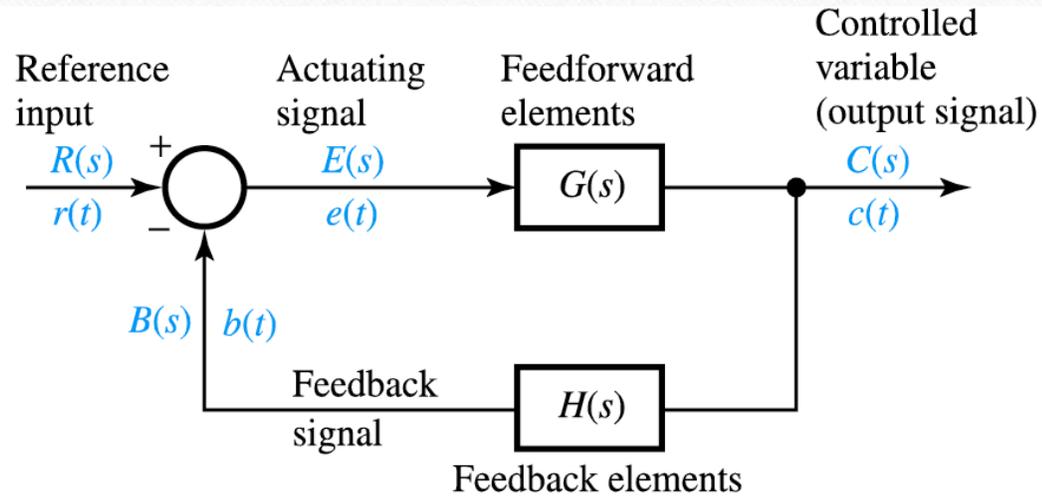
Mathematical Modeling



Component Block Diagram



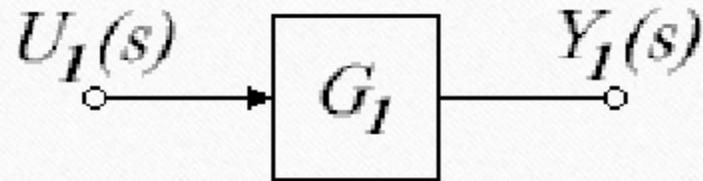
Component Block Diagram



- $R(s)$ Reference input
- $C(s)$ Output signal (controlled variable)
- $B(s)$ Feedback signal = $H(s)C(s)$
- $E(s)$ Actuating signal (error) = $[R(s) - B(s)]$
- $G(s)$ Forward path transfer function or open-loop transfer function = $C(s)/E(s)$
- $M(s)$ Closed-loop transfer function = $C(s)/R(s) = G(s)/[1 + G(s)H(s)]$
- $H(s)$ Feedback path transfer function
- $G(s)H(s)$ Loop gain
- $\frac{E(s)}{R(s)}$ = Error-response transfer function $\frac{1}{1 + G(s)H(s)}$

Component Block Diagram

- It represents the *mathematical relationships* between the elements of the system.



$$U_1(s) G_1(s) = Y_1(s)$$

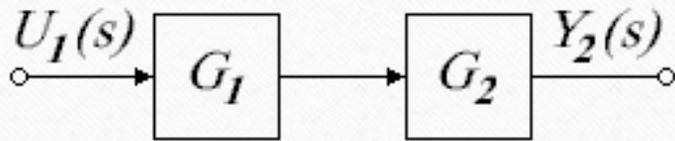
- The *transfer function* of each component is placed *in box*, and the *input-output relationships* between components are indicated by *lines and arrows*.

Component Block Diagram

- We can *solve the equations by graphical simplification*, which is often easier and more informative than algebraic manipulation, *even though the methods are in every way equivalent*.
- The interconnections of blocks include summing points, where any number of signals may be added together.

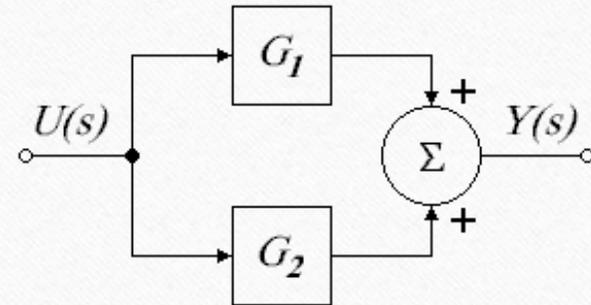
Block Diagram Reduction Technique

- Blocks in series:



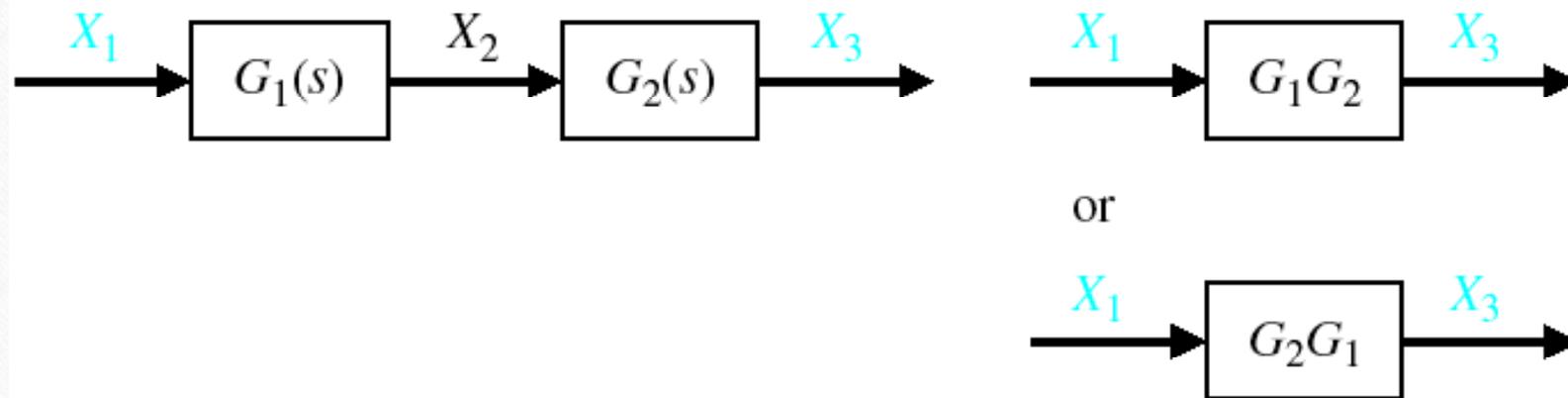
$$\frac{Y_2(s)}{U_1(s)} = G_1 G_2$$

- Blocks in parallel with their outputs added:



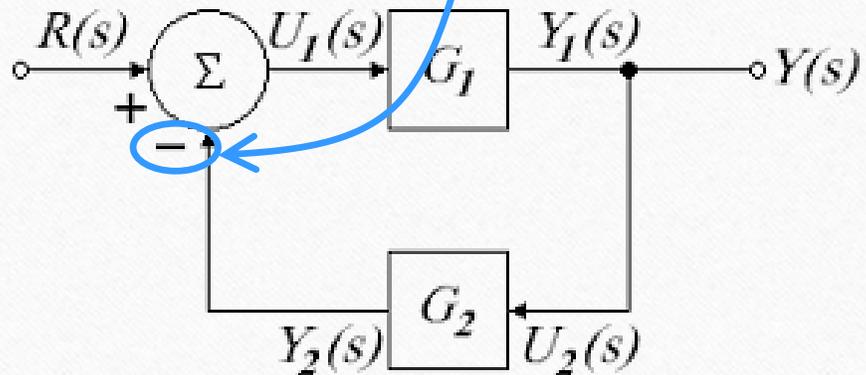
$$\frac{Y(s)}{U(s)} = G_1 + G_2$$

Combining blocks in cascade



Block Diagram Reduction Technique

- *Single-loop negative feedback*



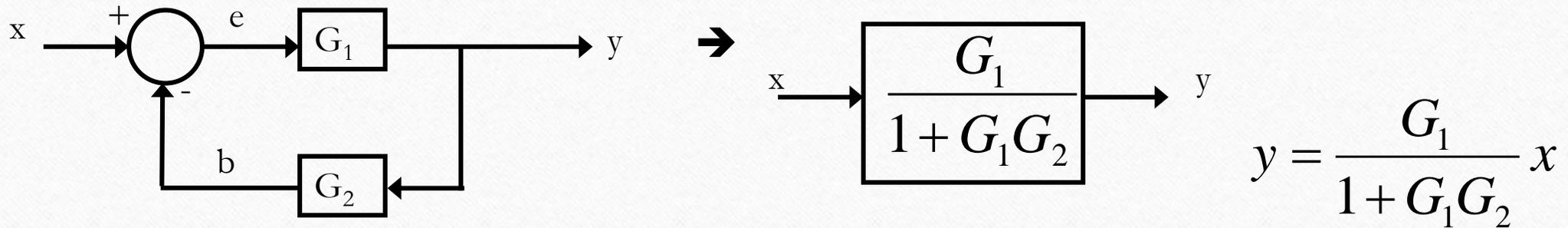
- Transfer function

$$\frac{Y(s)}{R(s)} = \frac{G_1}{1 + G_1 G_2}$$

Two blocks are connected in a feedback arrangement so that each feeds into the other.

Block Diagram Reduction Technique

- **Proof:**

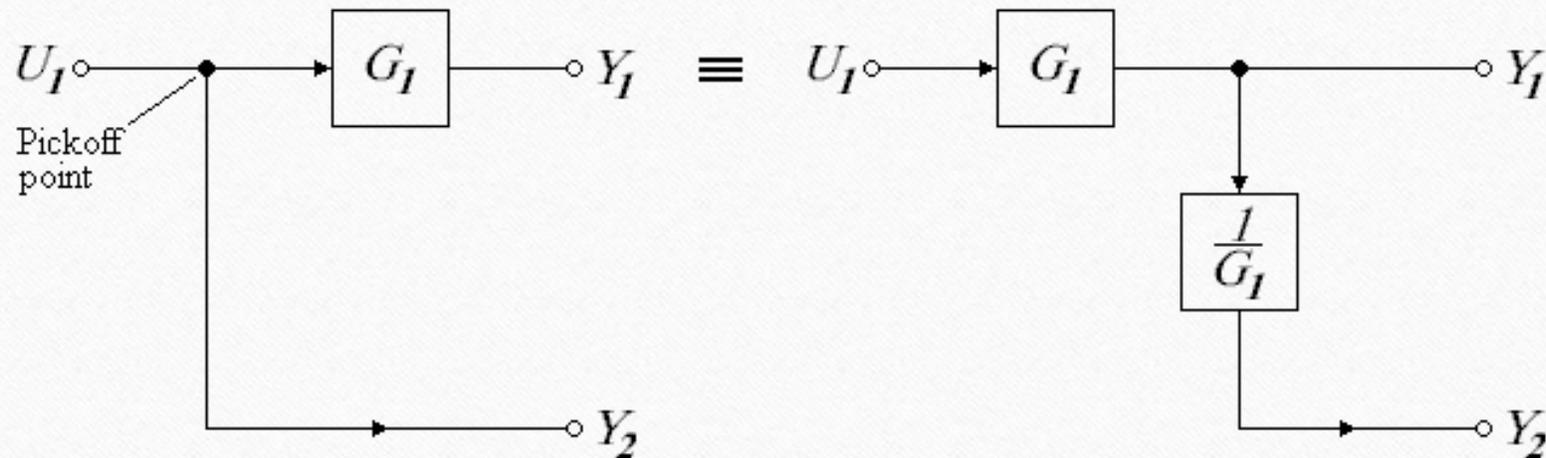


$$e = x - b, \quad b = G_2 y, \quad y = G_1 e \Rightarrow y = \frac{G_1}{1 + G_1 G_2} x$$

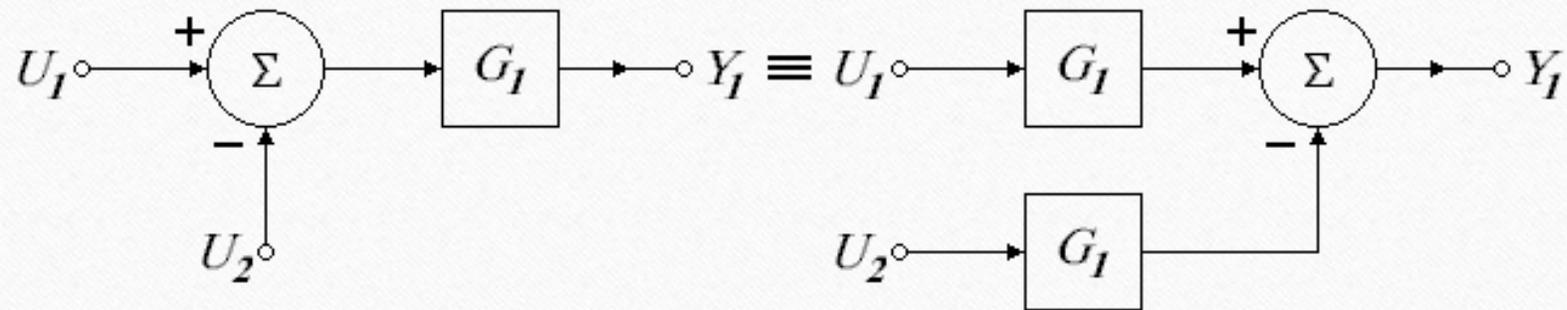
$$e = x - G_2 G_1 e$$

$$(1 + G_1 G_2) e = x \Rightarrow e = \frac{1}{1 + G_1 G_2} x$$

Block Diagram Reduction Technique



Block Diagram Reduction Technique



Block Diagram Reduction Technique

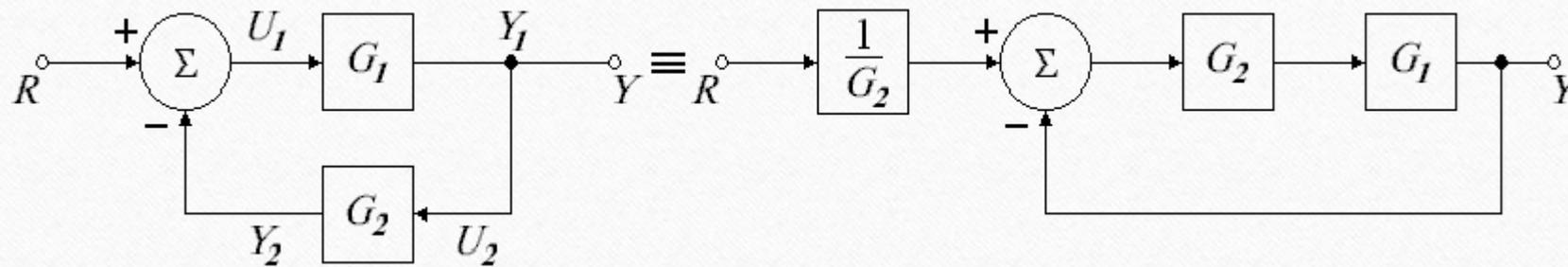
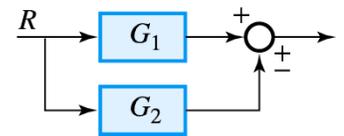
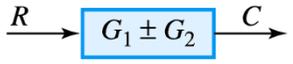
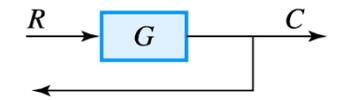
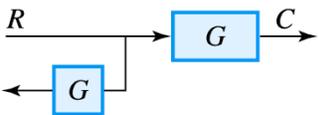
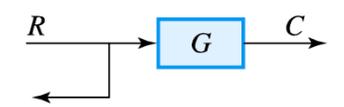
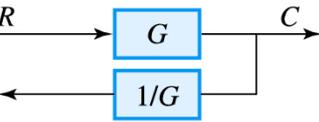
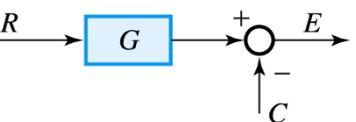
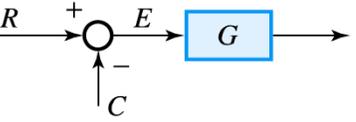
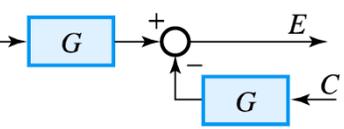
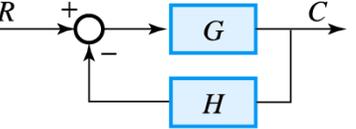
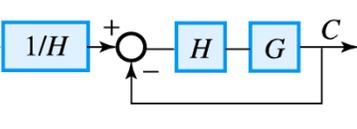
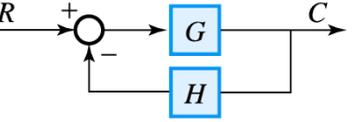
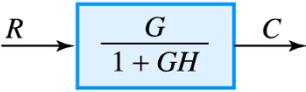


TABLE 3.4.1 Some of the Block Diagram Reduction Manipulations

| Original Block Diagram | Manipulation | Modified Block Diagram |
|--|--|---|
|  | Cascaded elements |  |
|  | Addition or subtraction (eliminating auxiliary forward path) |  |
|  | Shifting of pickoff point ahead of block |  |
|  | Shifting of pickoff point behind block |  |
|  | Shifting summing point ahead of block |  |
|  | Shifting summing point behind block |  |
|  | Removing H from feedback path |  |
|  | Eliminating feedback path |  |

Block Diagram Reduction Technique

Example

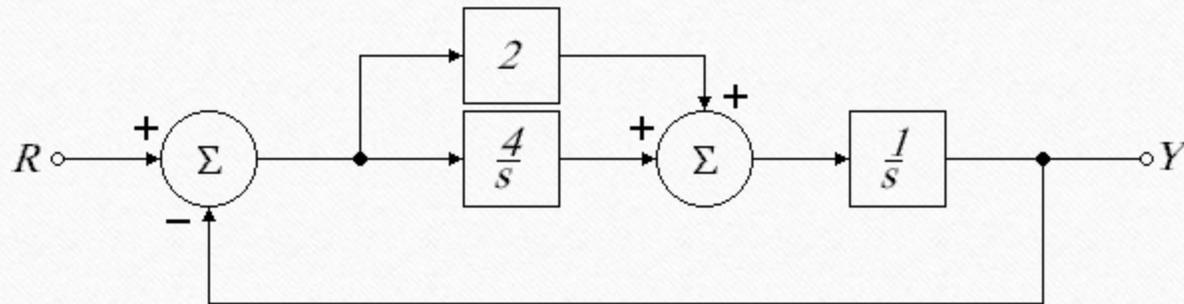


fig. (a)

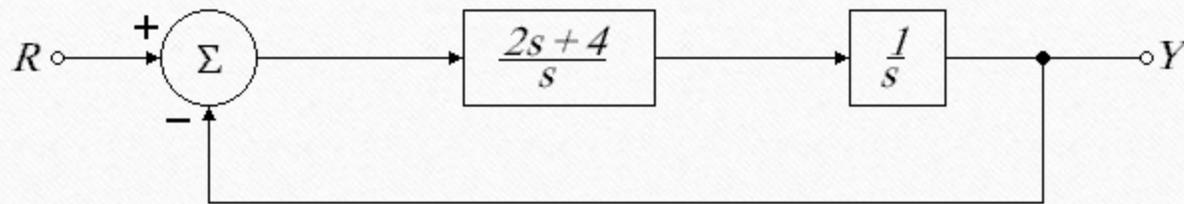


fig. (b)

$$T(s) = \frac{Y(s)}{R(s)}$$

$$T(s) = \frac{2s+4}{1 + \frac{s^2}{2s+4}}$$

$$T(s) = \frac{2s+4}{s^2 + 2s + 4}$$

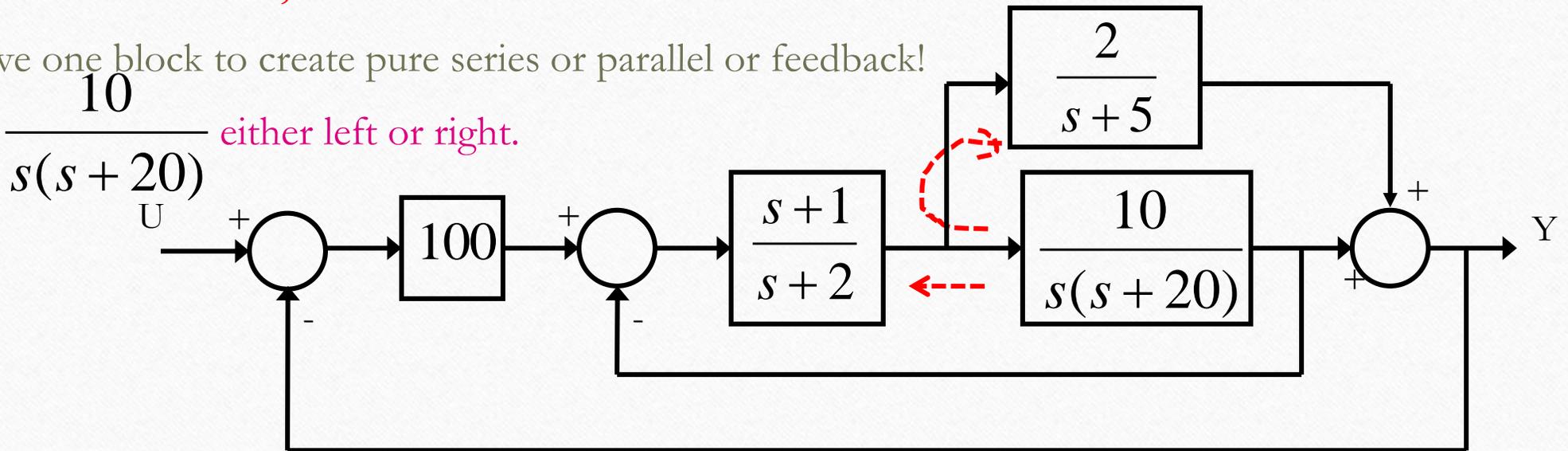
Block Diagram Reduction Technique

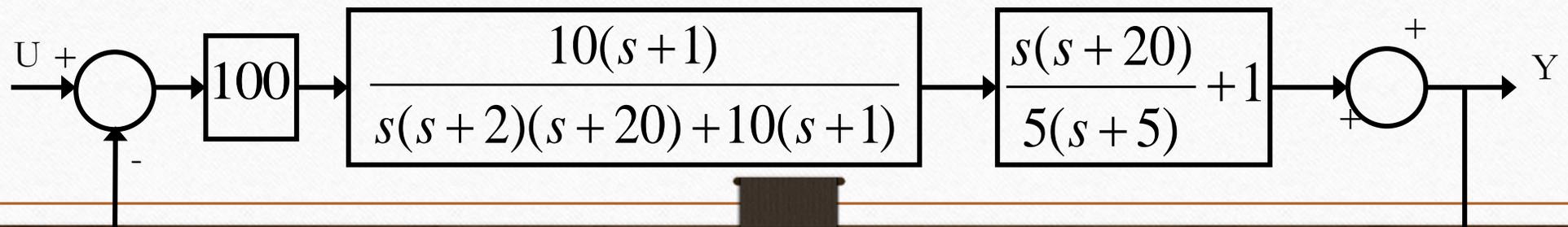
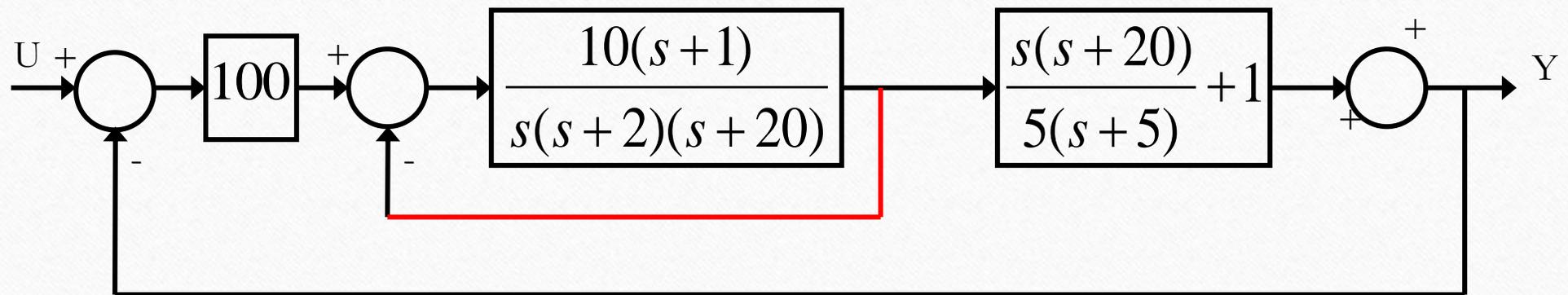
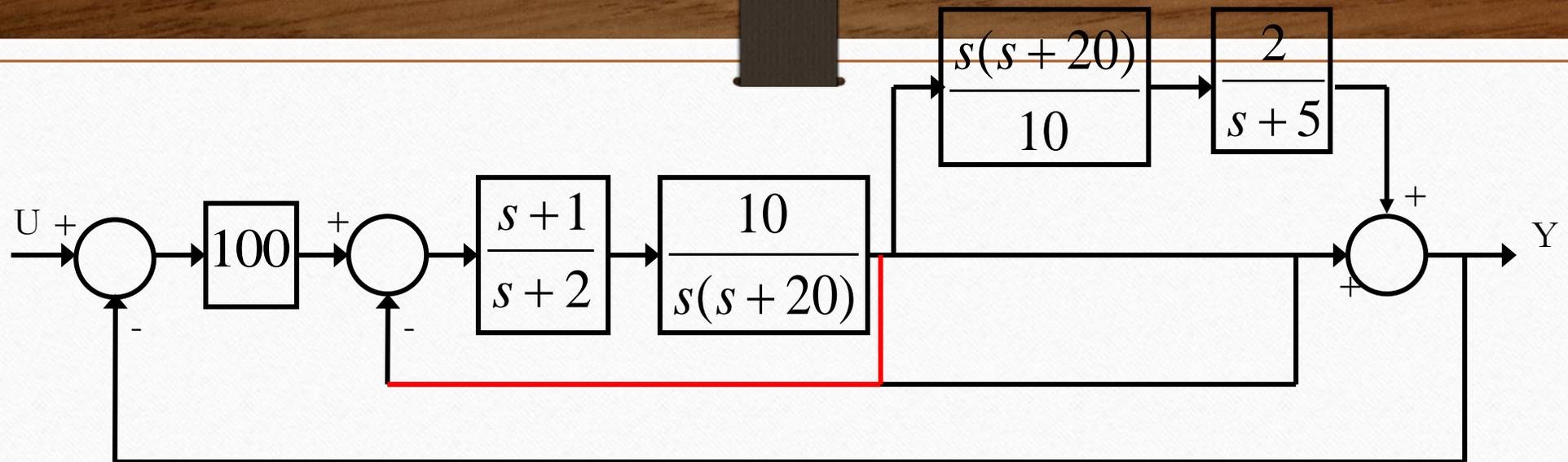
Example 2: Find TF from U to Y:

- No pure series/parallel/feedback
- Needs to move a block, but which one?

Key: move one block to create pure series or parallel or feedback!

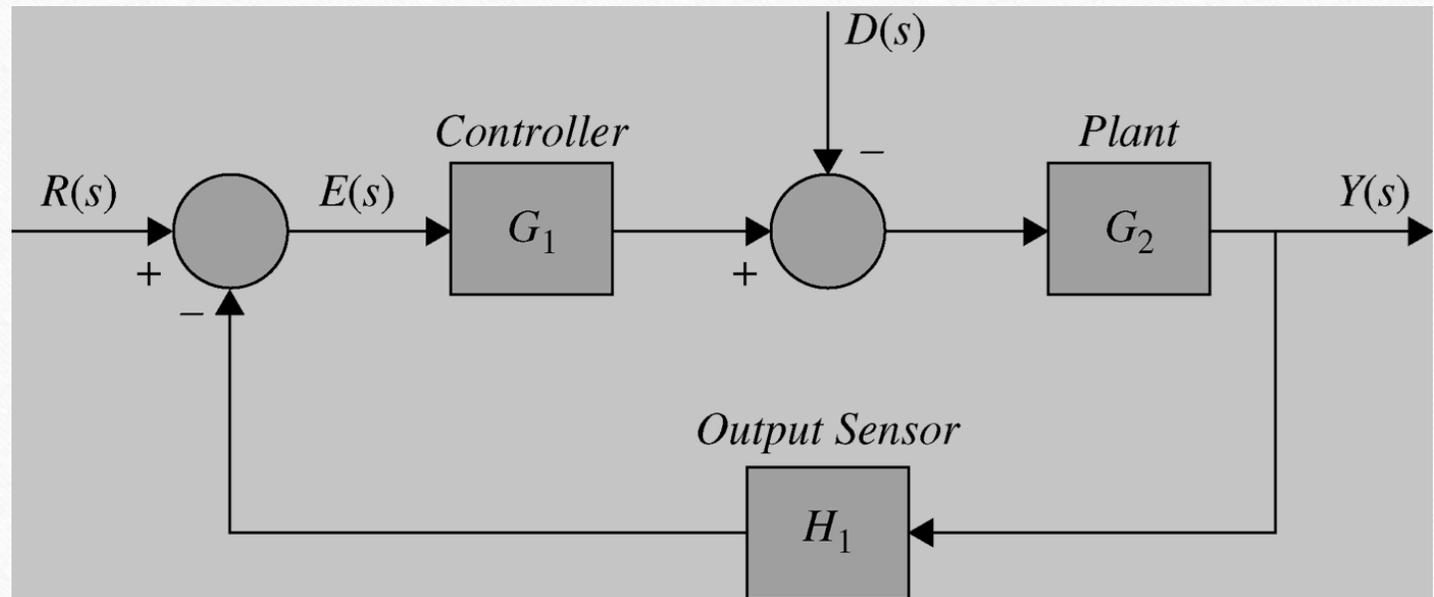
So move $\frac{10}{s(s+20)}$ either left or right.





Block Diagram Reduction Technique

Example



Can use superposition:

First set $D=0$, find Y due to R

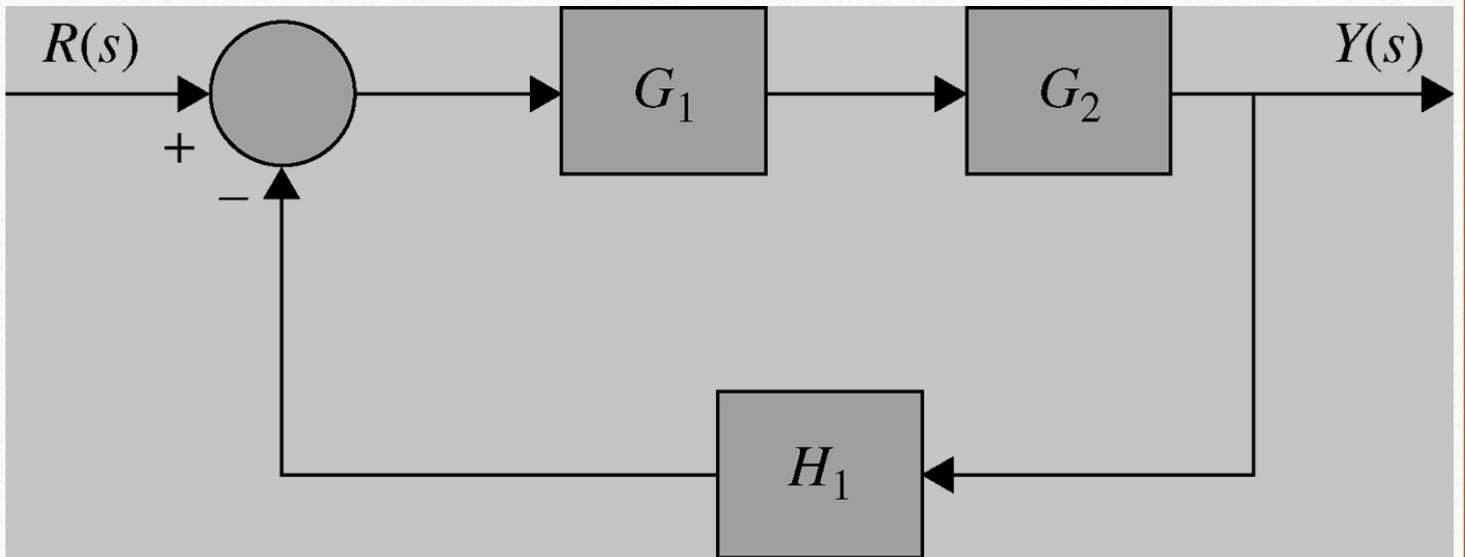
Then set $R=0$, find Y due to D

Finally, add the two component to get the overall Y

Block Diagram Reduction Technique

First set $D=0$, find Y due to R

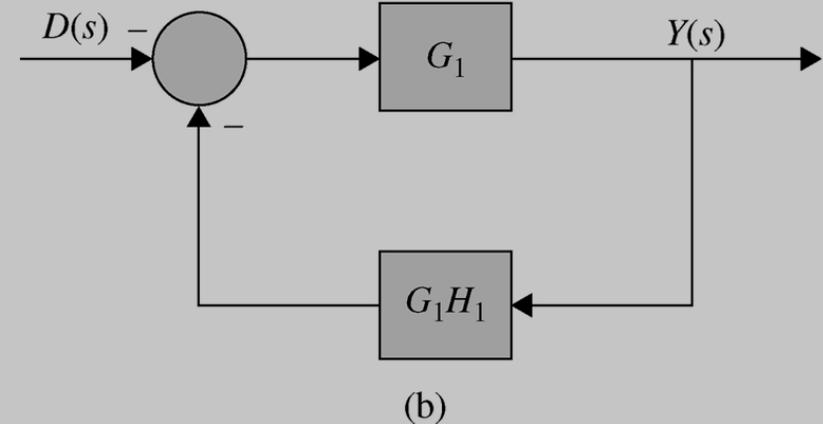
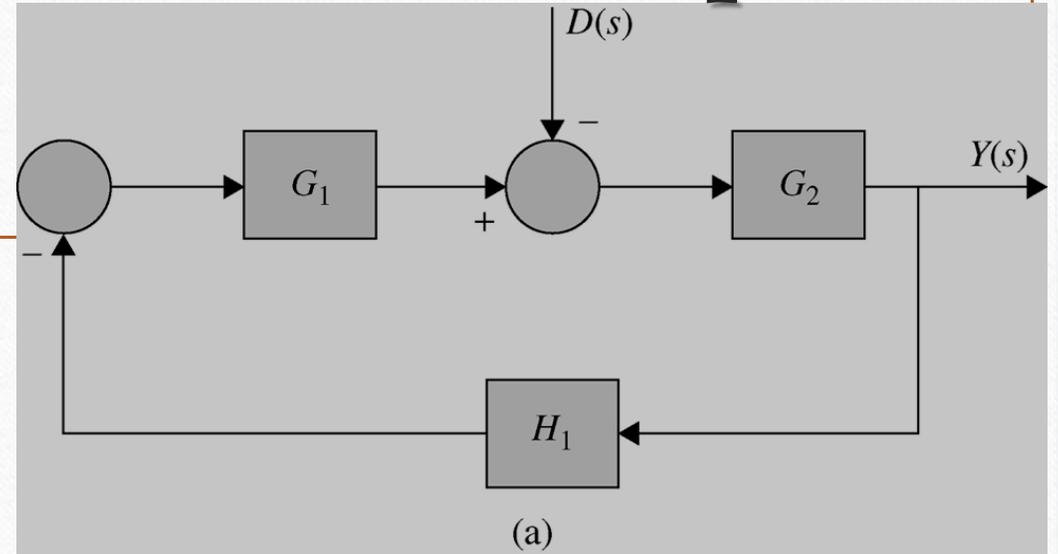
$$Y_1(s) = \frac{G_1 G_2}{1 + G_1 G_2 H_1} R(s)$$



Block Diagram Reduction Technique

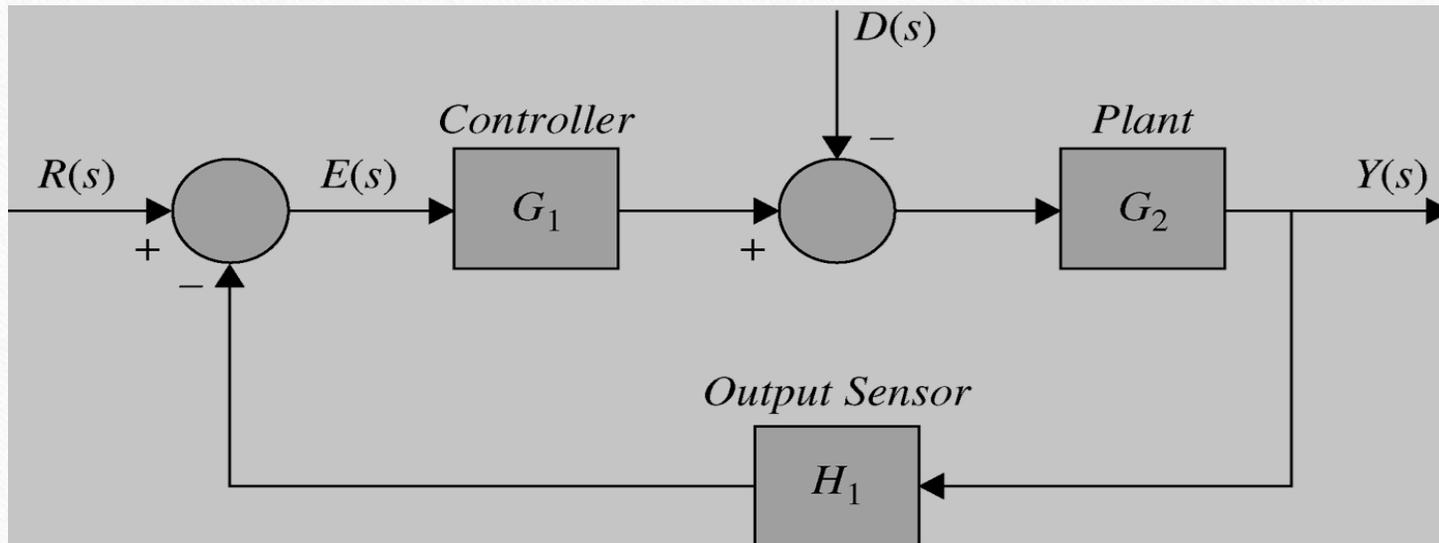
Then set $R=0$, find Y due to D

$$Y_2(s) = \frac{G_2}{1 + G_1 G_2 H_1} (-D(s))$$



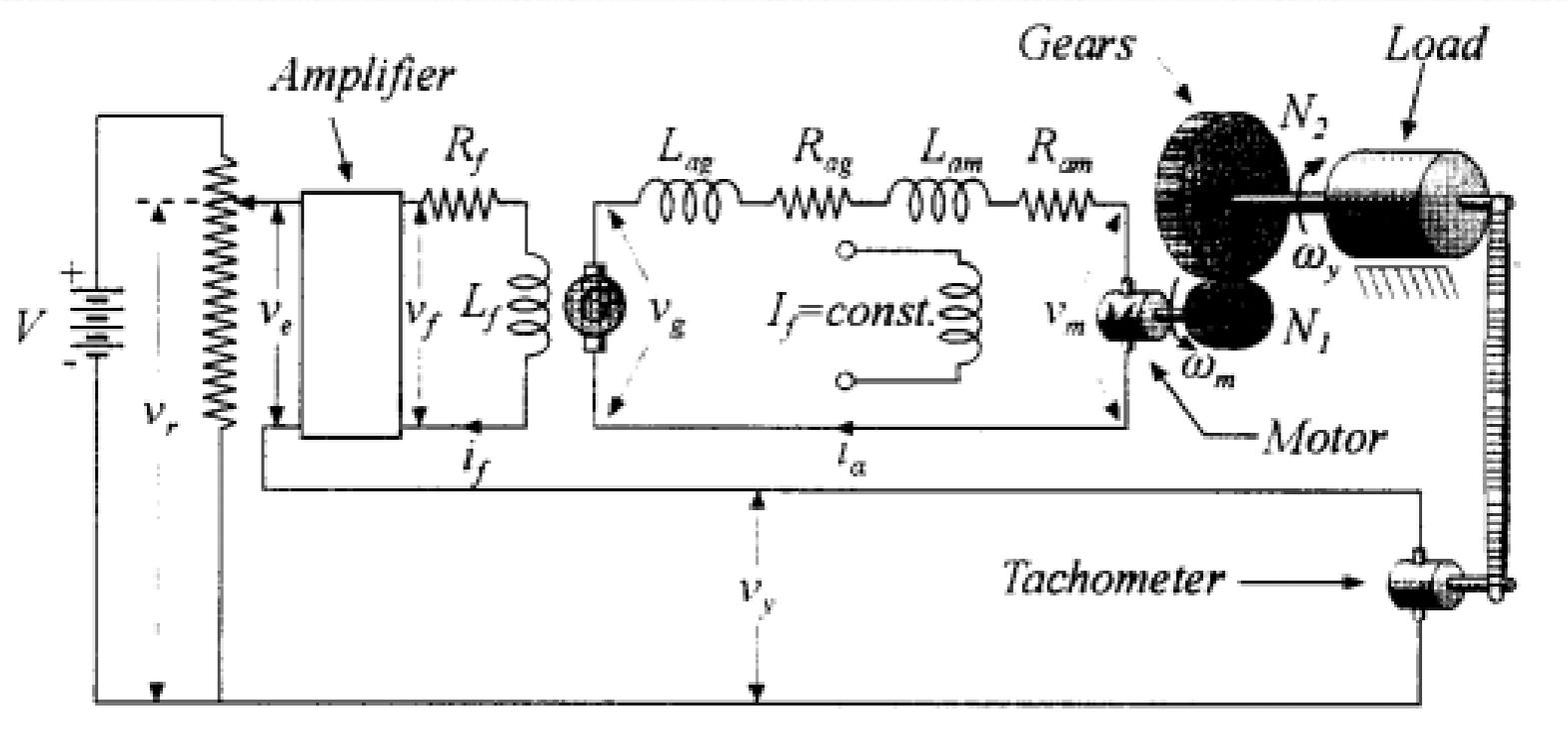
Block Diagram Reduction Technique

Finally, add the two component to get the overall Y



$$Y(s) = \frac{G_1 G_2}{1 + G_1 G_2 H_1} R(s) - \frac{G_2}{1 + G_1 G_2 H_1} D(s)$$

Modeling of Motors



Mathematical Modeling

The equations of the Ward–Leonard layout are as follows . The Kirchhoff's law of voltages of the excitation field of the generator G is

$$v_f = R_f i_f + L_f \frac{di_f}{dt}$$

The voltage v_g of the generator G is proportional to the current i_f , i.e.,

$$v_g = K_g i_f$$

The voltage v_m of the motor M is proportional to the angular velocity ω_m , i.e.,

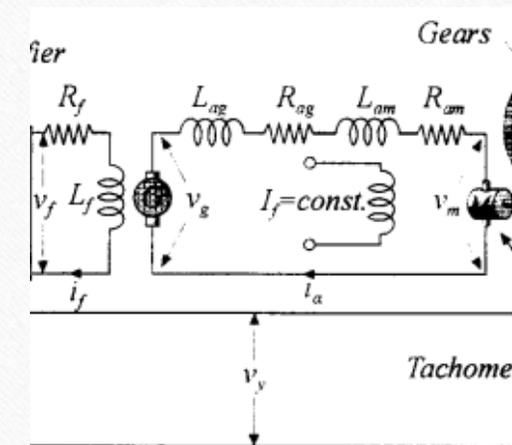
$$v_m = K_b \omega_m$$

The differential equation for the current i_a is

$$R_a i_a + L_a \frac{di_a}{dt} = v_g - v_m = K_g i_f - K_b \omega_m$$

The torque T_m of the motor is proportional to the current i_a

$$T_m = K_m i_a$$



Mathematical Modeling

The equations of the Ward–Leonard layout are as follows . The Kirchhoff's law of voltages of the excitation field of the generator G is

The rotational motion of the rotor is described by

$$J_m^* \frac{d\omega_m}{dt} + B_m^* \omega_m = K_m i_a$$

where $J_m^* = J_m + N^2 J_L$ and $B_m^* = B_m + N^2 B_L$, where $N = N_1/N_2$.

Here, J_m is the moment of inertia and B_m the viscosity coefficient of the motor: likewise, for J_L and B_L of the load.

where use was made of the relation

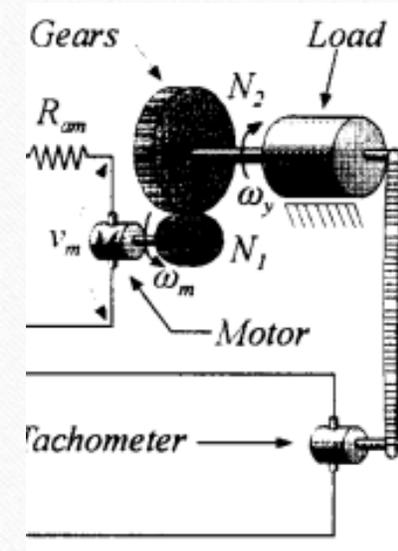
$$\omega_y = N \omega_m.$$

The tachometer equation

$$v_y = K_t \omega_y$$

the amplifier equation

$$v_f = K_a v_e$$

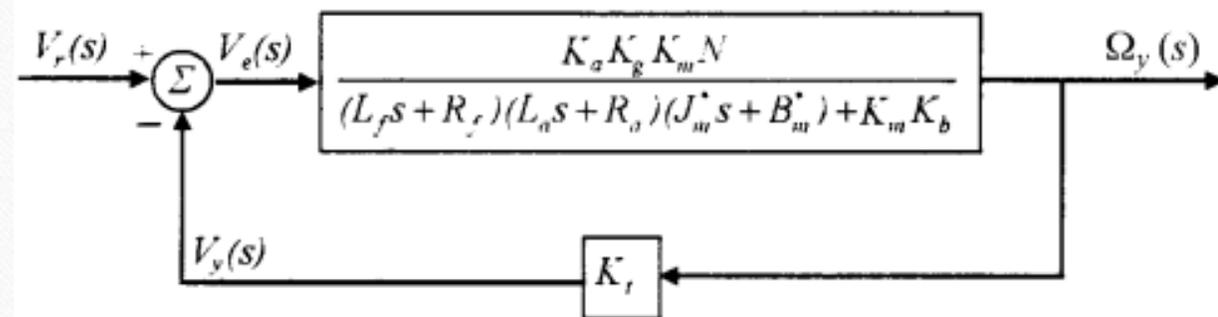


Mathematical Modeling

The mathematical model of the Ward–Leonard layout are as follows .

$$\frac{\Omega_y(s)}{V_f(s)} = \frac{K_g K_m N}{(L_f s + R_f)[(L_a s + R_a)(J_m^* s + B_m^*) + K_m K_b]}$$

$$\frac{\Omega_y(s)}{v_e(s)} = \frac{K_a K_g K_m N}{(L_f s + R_f)[(L_a s + R_a)(J_m^* s + B_m^*) + K_m K_b]}$$



The state variables of a dynamic system

Input signals

$u_1(t)$



$u_2(t)$



System

Output signals

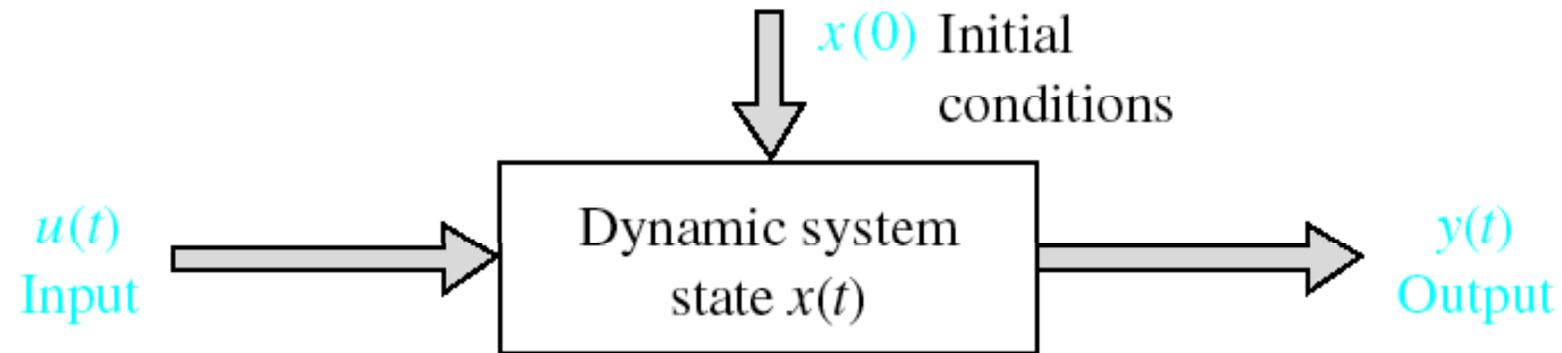
$y_1(t)$



$y_2(t)$



The general form of a dynamic system



State Space Equations

- **State equations** is a description which relates the following four elements: input, system, state variables, and output

$$\dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) + \mathbf{B}\mathbf{u}(t)$$

$$\mathbf{y}(t) = \mathbf{C}\mathbf{x}(t) + \mathbf{D}\mathbf{u}(t)$$

Matrix A has dimensions $n \times n$ and it is called the **system** matrix, having the general form

$$\mathbf{A} = \begin{bmatrix} a_{11} & a_{12} & \cdots & a_{1n} \\ a_{21} & a_{22} & \cdots & a_{2n} \\ \vdots & \vdots & & \vdots \\ a_{n1} & a_{n2} & \cdots & a_{nn} \end{bmatrix}$$

Matrix B has dimensions $n \times m$ and it is called the **input** matrix, having the general form

$$\mathbf{B} = \begin{bmatrix} b_{11} & b_{12} & \cdots & b_{1m} \\ b_{21} & b_{22} & \cdots & b_{2m} \\ \vdots & \vdots & & \vdots \\ b_{n1} & b_{n2} & \cdots & b_{nm} \end{bmatrix}$$

Matrix C has dimensions $p \times n$ and it is called the **output** matrix, having the general form

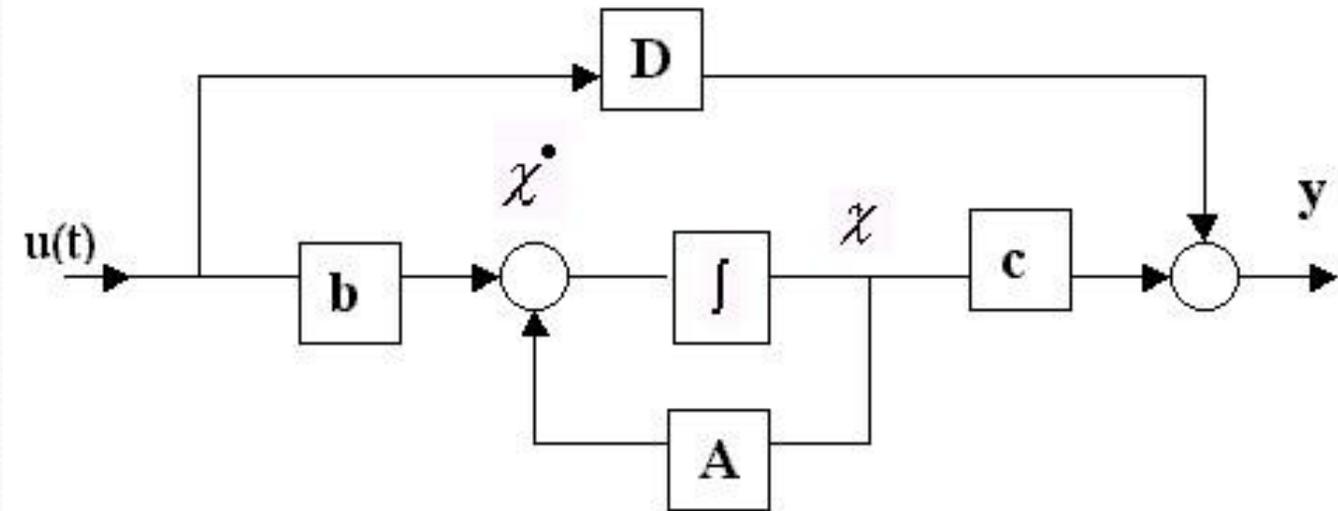
$$\mathbf{C} = \begin{bmatrix} c_{11} & c_{12} & \cdots & c_{1n} \\ c_{21} & c_{22} & \cdots & c_{2n} \\ \vdots & \vdots & & \vdots \\ c_{p1} & c_{p2} & \cdots & c_{pn} \end{bmatrix}$$

Matrix D has dimensions $p \times m$ and it is called the **feedforward** matrix, having the general form

$$\mathbf{D} = \begin{bmatrix} d_{11} & d_{12} & \cdots & d_{1m} \\ d_{21} & d_{22} & \cdots & d_{2m} \\ \vdots & \vdots & & \vdots \\ d_{p1} & d_{p2} & \cdots & d_{pm} \end{bmatrix}$$

State Space Equations

$$\begin{aligned} \text{SISO} \Rightarrow \dot{X} &= Ax(t) + Bu(t) \\ Y &= Cx(t) + Du(t) \end{aligned}$$



State Space representation

- **The general form of a dynamic system**

The concept of a set of state variables that represent a dynamic system can be illustrated in terms of the spring-mass-damper system. A set of state variables sufficient to describe this system includes the position and the velocity of the mass.

- We will define a set of state variables as (x_1, x_2) , where

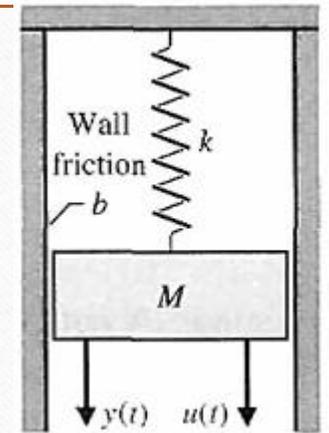
$$x_1(t) = y(t) \quad \text{and} \quad x_2(t) = \frac{dy(t)}{dt} \quad \frac{dx_1}{dt} = x_2$$

To write Equation of motion in terms of the state variables, we substitute the state variables as already defined and obtain

$$M \frac{dx_2}{dt} + bx_2 + kx_1 = u(t)$$

Therefore, we can write the equations that describe the behavior of the spring-mass damper system as the set of two first-order differential equations

$$\frac{dx_2}{dt} = \frac{-b}{M}x_2 - \frac{k}{M}x_1 + \frac{1}{M}u$$



$$M \frac{d^2y}{dt^2} + b \frac{dy}{dt} + ky = u(t)$$

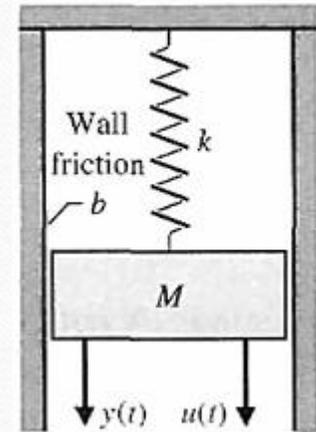
State Space representation

- State space matrix

$$\frac{dx_1}{dt} = x_2$$

$$\frac{dx_2}{dt} = \frac{-b}{M}x_2 - \frac{k}{M}x_1 + \frac{1}{M}u$$

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ \frac{-k}{m} & \frac{-b}{m} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ \frac{1}{m} \end{bmatrix} u(t)$$



$$M \frac{d^2y}{dt^2} + b \frac{dy}{dt} + ky = u(t)$$

State Space representation

- Transfer from time domain to frequency domain:

$$R_1 i_1(t) + \frac{1}{C} \int_0^t i_1(t) dt - \frac{1}{C} \int_0^t i_2(t) dt = v(t)$$

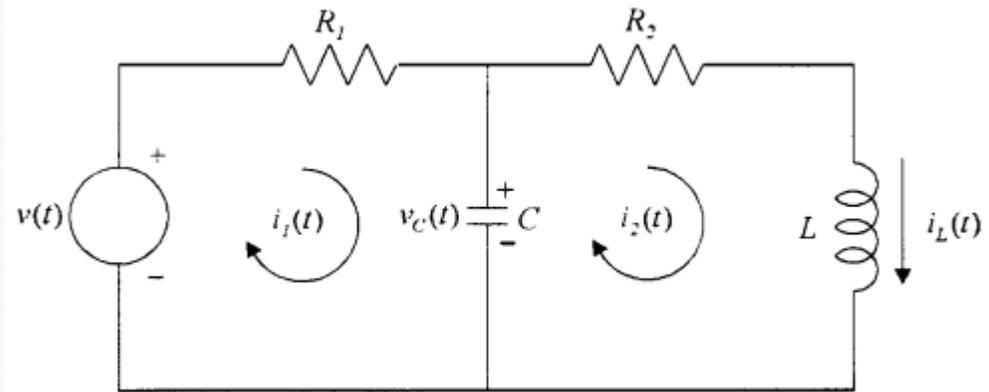
$$\left[R_1 + \frac{1}{Cs} \right] I_1(s) - \frac{1}{Cs} I_2(s) = V(s)$$

$$-\frac{1}{C} \int_0^t i_1(t) dt + R_2 i_2(t) + L \frac{di_2}{dt} + \frac{1}{C} \int_0^t i_2(t) dt = 0$$

$$-\frac{1}{Cs} I_1(s) + \left[R_2 + Ls + \frac{1}{Cs} \right] I_2(s) = 0$$

- Transfer function

$$\frac{I_2(s)}{V(s)} = \frac{Cs}{(R_1 Cs + 1)(LCs^2 + R_2 Cs + 1) - 1} = \frac{1}{R_1 LCs^2 + (R_1 R_2 C + L)s + R_1 + R_2}$$



State Space representation

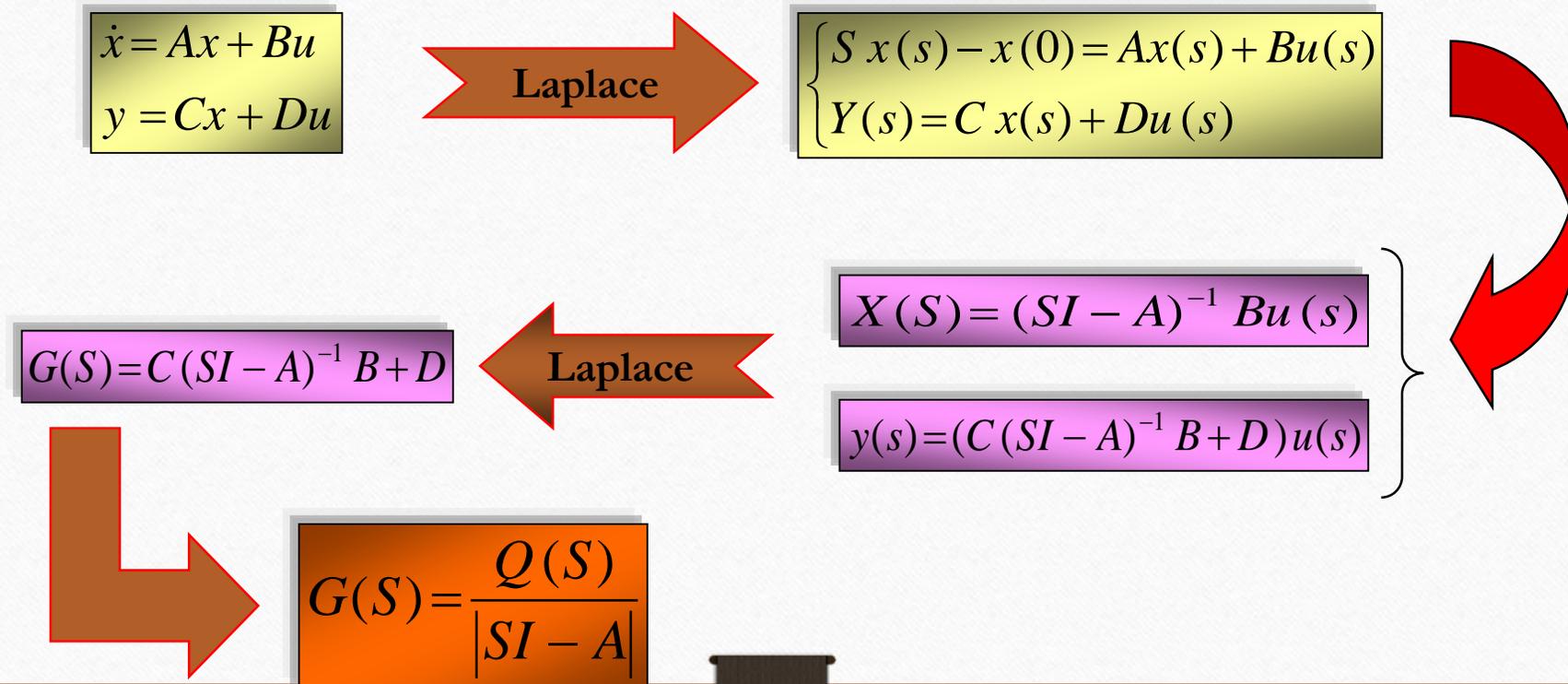
$$\begin{cases} e(t) - R_1 i_1(t) - L_1 \frac{di_1}{dt} - V_C(t) = \phi \\ V_C(t) - L_2 \frac{di_2}{dt} - R_2 i_2 = \phi \\ i_c = i_1 - i_2 = C \frac{dv_c}{dt} \end{cases}$$

$$x = (i_1 \quad i_2 \quad v_c)^T$$

$$\dot{X} = \begin{pmatrix} \frac{-R_1}{L_1} & 0 & \frac{-1}{L_1} \\ 0 & \frac{-R_2}{L_2} & \frac{1}{L_2} \\ \frac{1}{C} & \frac{-1}{C} & 0 \end{pmatrix} \begin{pmatrix} x_1 \\ x_2 \\ x_3 \end{pmatrix} + \begin{pmatrix} \frac{1}{L_1} \\ 0 \\ 0 \end{pmatrix} e(t)$$

$$y(t) = (0 \quad R_2 \quad 0) \begin{pmatrix} x_1 \\ x_2 \\ x_3 \end{pmatrix}$$

State Space representation



State Space representation

$$\begin{cases} \dot{x}_1 = -5x_1 - x_2 + 2u \\ \dot{x}_2 = 3x_1 - x_2 + 5u \end{cases} \Rightarrow \dot{x} = \begin{pmatrix} -5 & -1 \\ 3 & -1 \end{pmatrix} x + \begin{pmatrix} 2 \\ 5 \end{pmatrix} u$$

$$y = x_1 + 2x_2$$

$$y = (1 \quad 2) x$$

$$(SI - A) = \begin{pmatrix} 2+5 & 1 \\ -3 & S+1 \end{pmatrix}$$

$$(SI - A)^{-1} = \frac{1}{\underbrace{(S+5)(S+1)+3}_{\Delta=(S+2)(S+4)}} \begin{pmatrix} S+1 & -1 \\ 3 & S+5 \end{pmatrix}$$

$$G(S) = [1 \quad 2] \frac{1}{\Delta} \begin{bmatrix} S+1 & -1 \\ 3 & S+5 \end{bmatrix} \begin{bmatrix} 2 \\ 5 \end{bmatrix}$$

$$G(s) = \frac{12S + 59}{(S+2)(S+4)}$$

Model Examples

Quadrocopter Pole Acrobatics





Thank you